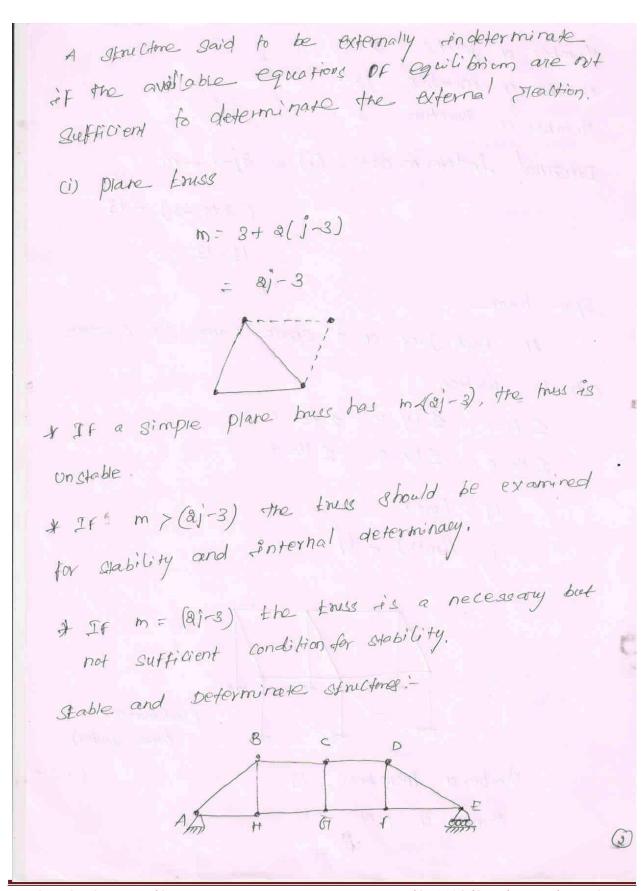
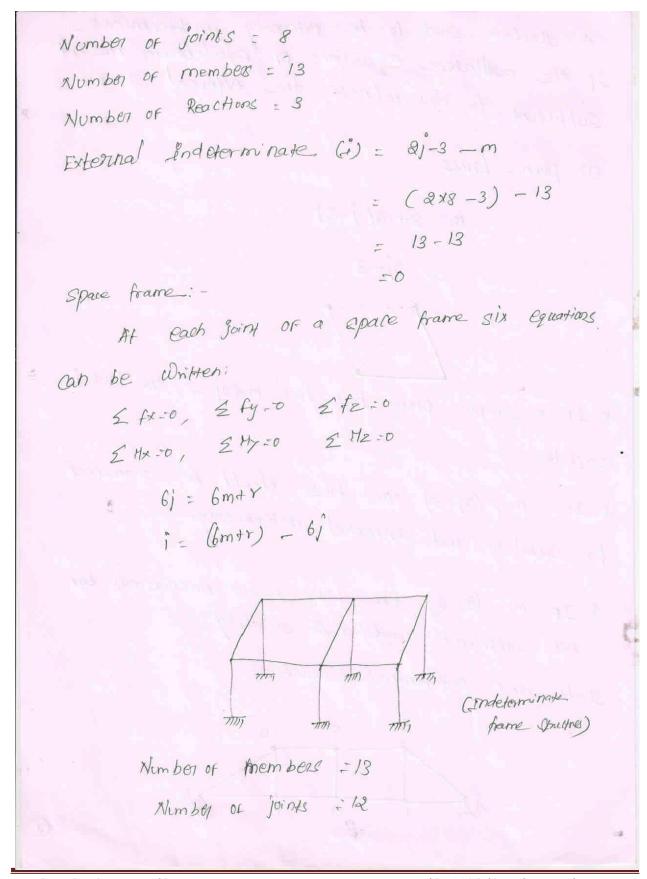
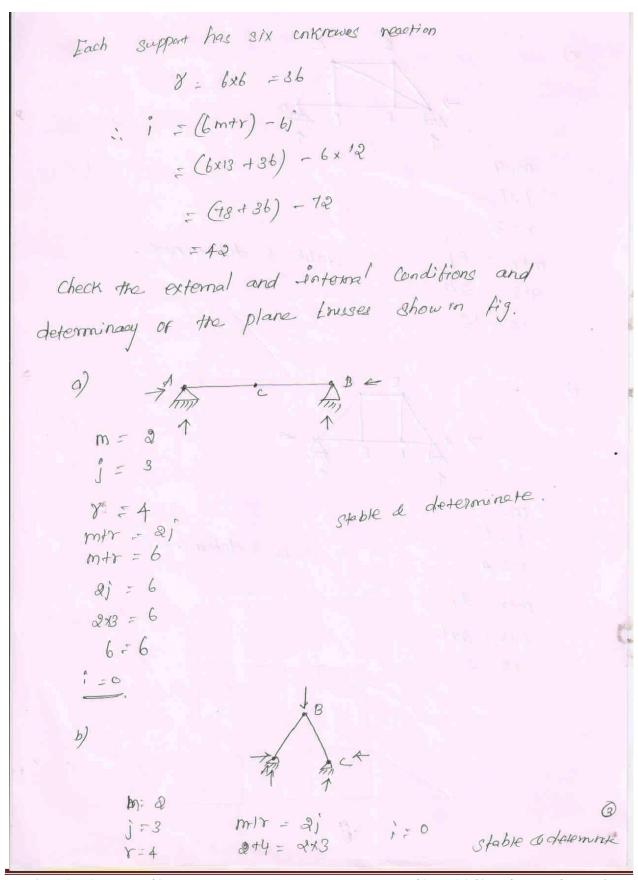
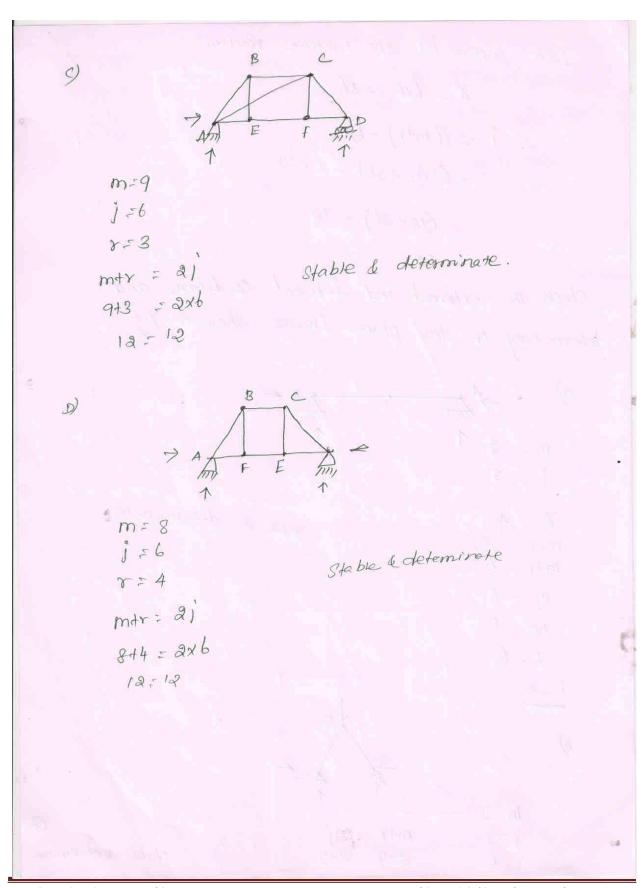
## UNIT-I FLEXIBILITY MATRIX METHOD Equilibrium and compatibility - Determinate Vs indeterminate Astruitures - indeterminacy - primary structure - competibling Conditions - Analysis of indeterminate pin jointed plane Frames, continuous Beams, rigid plane frames. Inbroduction A structure may be get to be in Thet con in motion. The usual Civic Engineering Structures Say, a building, a dam (on a bridge is all set at stept and hence may be be said to be in statical equilibrium. An in flight alloplane on a moving train is in. motion and hence is in dynamic equilibrium. Static Equilibrium: The equation of static equilibrium are based on six isacc Newton's law governing the motions of boolies which say (a) The Scm of all forces in any axis 15 zon le 2 fx=0; & fy=0; & F2=0 (6) The Som of all the moments about any axis is - E Mx=0; E My=0; and E Mz=0 0

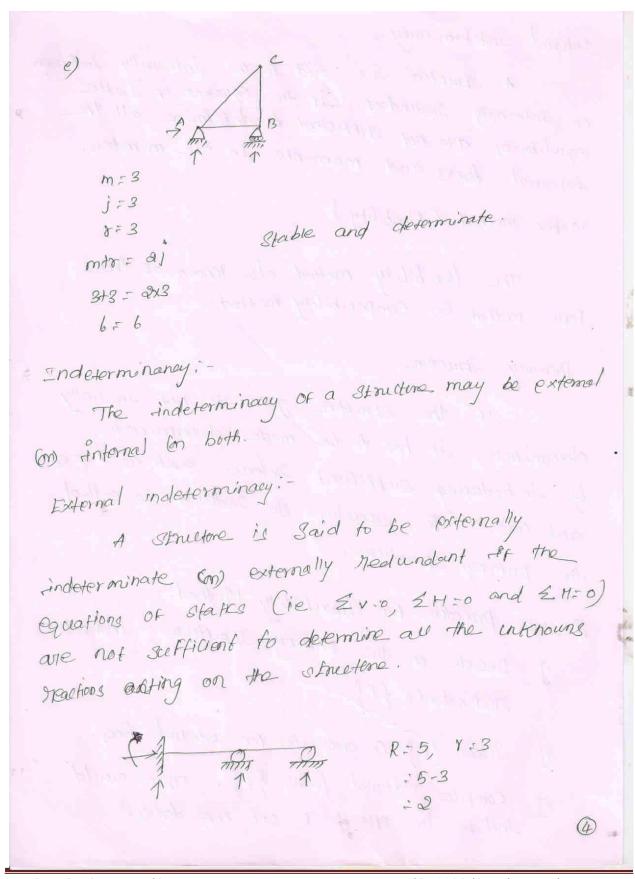
In general, six independent Condition of equilibrium Compati bility : exists. They are (a) Sum of the forces in three crithogonal directions is zero. (b) Sum of moments about the three, orthogonal axes is toro. Determinate VS indeterminate Structures: \* A SENICEONE is said to be statically determinate if the equations of state equilibrium Cie St-0 and EM.0) are gufficient to determinate all the internal forces and moments in the members. A Structure is said to statically indeterminate if the equation of Static equilibrium over consider to determine all the faces and moments in the General Criteria For determining stability Determinancy members. and indeterminary of a spriteral members: very broadly, a planar smulture may be a beam, a frame or a fruit. A spectures may be externally in determinate on internally rinderestminate.











Internal indeterminacy A structure is said to be internally indeterminate or internally medendant, if the equations of static equilibrium are not gufficient to determine all the internal forces and moments in the members. Makix Method (Flexibility) The Aexibility method also known of the force method to compatibility method. Primary Structure If the structure given is not initially aletominate, it has to be made determinate, by in froducing sufficient nelesses such as higger and cuts. The structure so Hedund is called The primary structure. 8 Steps procedure For Flexibility Method: if Decide on the primary structures; indicate Medundants of f. 2] Select IPS Co-ordinates for interior fines. 3] compile external forces of fg. This would include fre effects of off node forces.

Analyse the Continuous beam in Aig. by

Flexibility method:

16 km

A Jammortonom B

FF = 3 min, EF=1 min, Example 1 Solution No. of unknowns = 5 Avilable ean = 3 degree of gredendary = 5-3

The sme two has 2 degree of reductiony. The primary structure is chosen as below, by introducing hinges at A and B. The moments at A and B are that needendants. 63° F4° C please not that f4 is a pain of moments making a bending moment.

Step:2  The fpf system is  Flement	as under  3  4  or co-ordinates
elateral loads on AB and Barried and moments.	is in 2 parts. If and It of the are opposited at a the opposited at a contract of an area of the opposited at a contract of an area of a contract of
// A	Name of moment Equivalent joints loools
	- 33
BC - wl = -20.0 +wl = +20.0	M8C 20 0 MCB -20

Hence 
$$f_{i}^{*} = 413.30 \text{ kW/m}$$
 $f_{i}^{*} = -20.0 \text{ kW/m}$ 

The EbJ matrix is generated by introducing

 $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$  and  $f_{i}^{*} = 1$  in steps

 $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$  and  $f_{i}^{*} = 1$  in steps

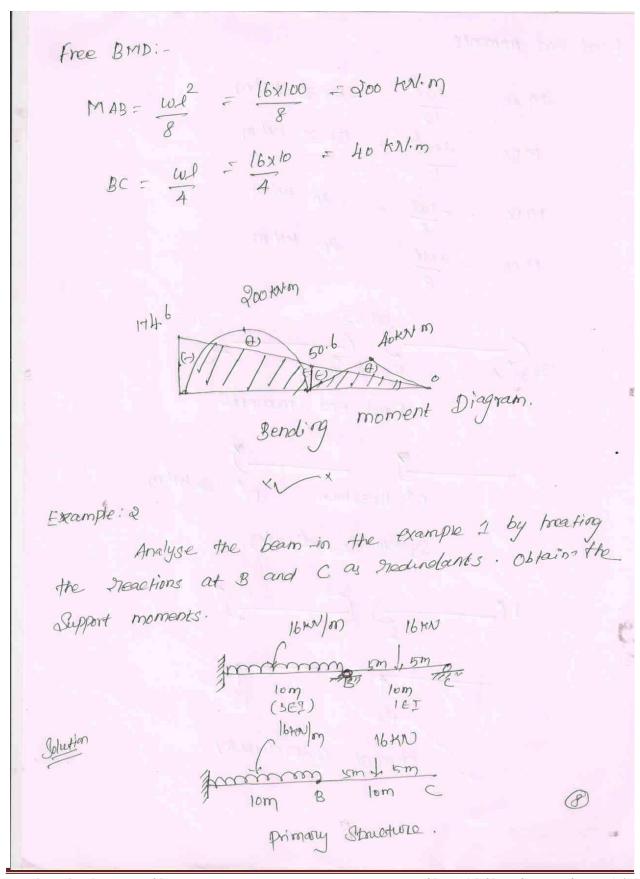
and finding  $f_{i}^{*}$ ,  $f_{i}^{*}$ ,  $f_{i}^{*}$ , and  $f_{i}^{*}$ .

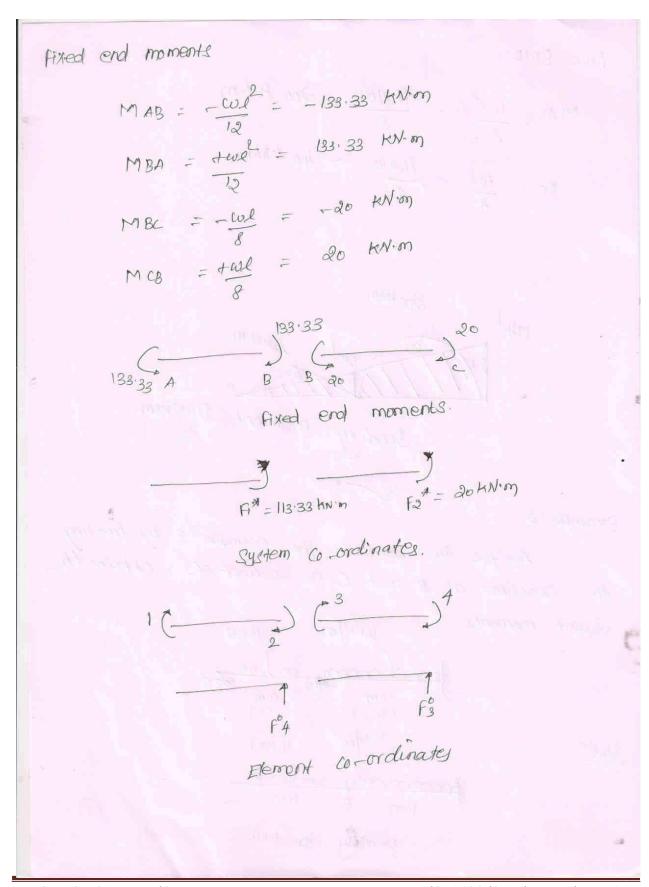
Hence  $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$  and  $f_{i}^{*} = 1$  in steps

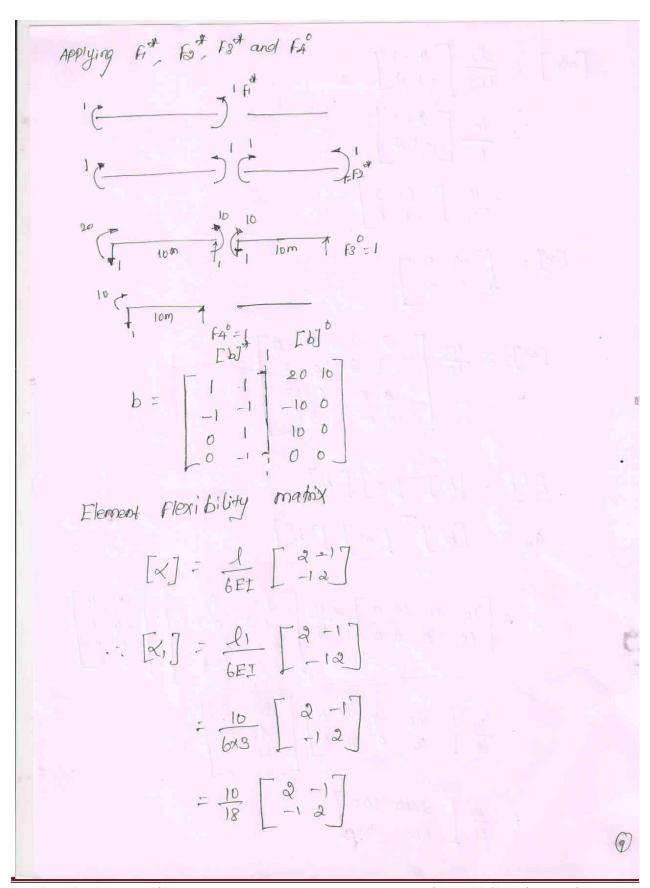
 $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$ ,  $f_{i}^{*} = 1$  and  $f_{i}^{*} = 1$  in steps

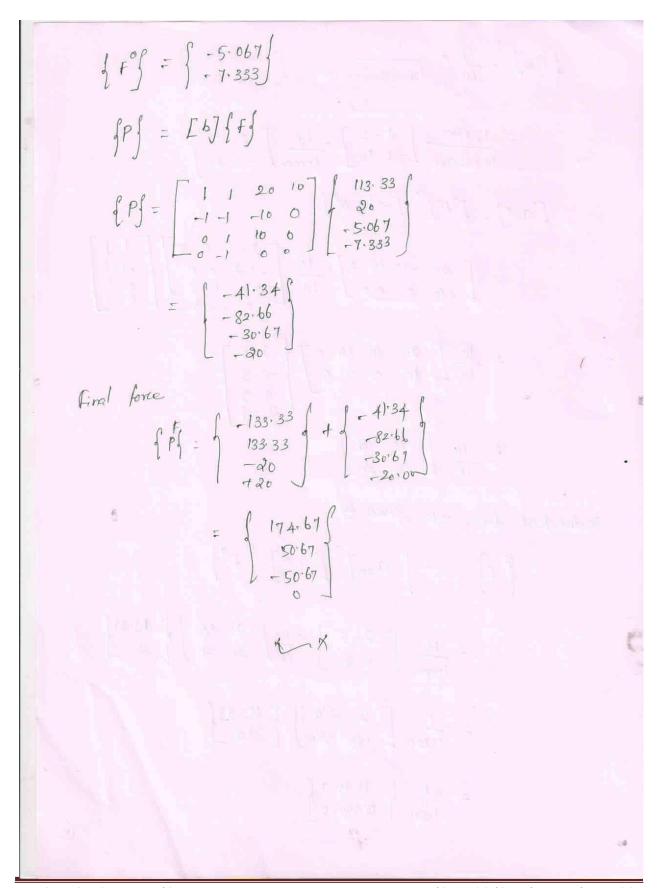
 $f_{i}^{*} = 1$ ,  $f_{i}^{*$ 

$$\begin{aligned}
& \{f\}^{\circ} = -\left[a_{00}\right]^{\circ} \left[a_{0}^{*}\right] \{f\}^{\circ} \\
& = o(12) \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right] \{f\}^{\circ} \\
& = \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right] \{f\}^{\circ} \\
& = \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right] \{f\}^{\circ} \\
& = \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right]^{-1} \left[a_{0}^{*}\right]^{-1} \\
& = \left[a_{0}^{*}\right]^{-1} \left[a_{0}^$$

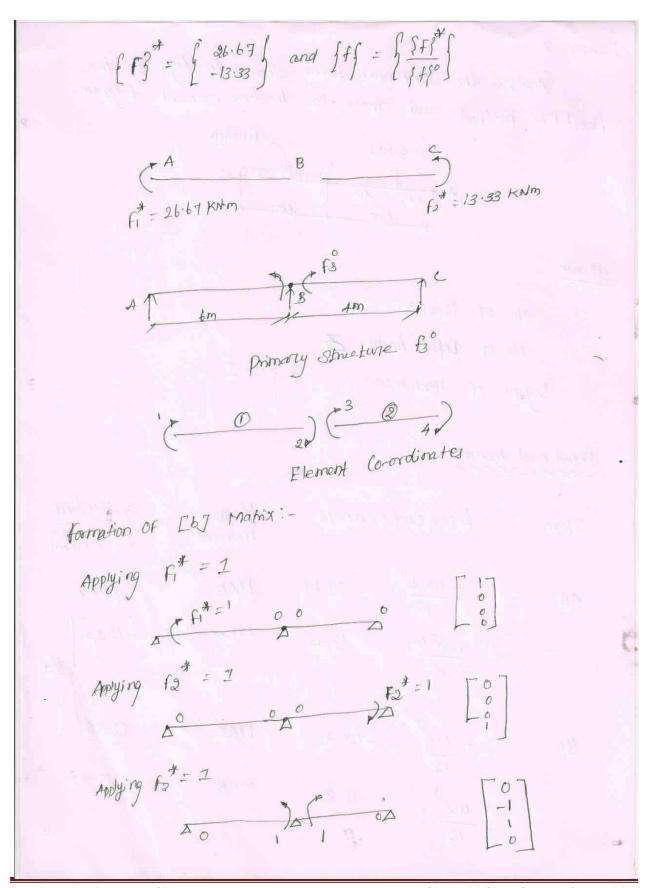








Solution: - No. No.	e the Continuous Beam  method and chaw the  sokn  and Am  am  am  am  am  am  am  am  am  am	m c	By the diagram.
Span	fixed end moments	Name of moment	Equilatent joint force (kN on)
AB	- wab = -26-67	МАВ	26.67
ВС	$\frac{\omega a^2 b}{\ell^2} = 13.33$ $-\omega \ell^2 = -13.33$	MBA MBC	13.33
	12 wl <sup>2</sup> = 13.33	MCB	- 13·33 6)



Hence 
$$\Gamma b J = Matrix$$
:

$$\Gamma b J = \begin{bmatrix} 1 & 0 & | & 0 \\ 0 & 0 & | & -1 \\ 0 & 0 & | & +1 \end{bmatrix}$$

Flement Flexibility Matrix  $\Gamma a J J = \begin{bmatrix} 1 & 2 & -1 \\ -1 & 2 \end{bmatrix}$ 

$$= \begin{bmatrix} 1 & 2 & -1 \\ 6EI & -1 & 2 \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 2 & -1 \\ -1 & 2 \end{bmatrix}$$

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$$= \begin{bmatrix} 1 & 2 & -1 & 0 \\ -1 & 2 & 0 & 0 \\ -1 & 2 & 0 & 0 \\ -1 & 2 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 1 & 2 & -1 \\ -1 & 2 & 3 \end{bmatrix}$$
To Find Redundant Forces
$$\begin{cases} \Gamma f = - \Gamma a^{0} J \Gamma a \Gamma \Gamma a \Gamma B J \Gamma B J$$

$$\begin{bmatrix} a_{00} \end{bmatrix} = \begin{bmatrix} 0 & -1 & 1 & 0 \end{bmatrix} \times \frac{1}{EI} \begin{bmatrix} 2 & -1 & 0 & 0 \\ -1 & 2 & 0 & 0 \\ 0 & 0 & 1/33 & -0/67 \end{bmatrix} \begin{bmatrix} 0 \\ -1 \\ 0 \end{bmatrix}$$

$$= \frac{1}{EI} \begin{bmatrix} 1 - 2 & 1/33 & -0/67 \end{bmatrix} \begin{bmatrix} 0 \\ -1 \\ 0 \end{bmatrix}$$

$$= \frac{1}{EI} \begin{bmatrix} 3.33 \end{bmatrix}$$

$$\begin{bmatrix} a_{00} \end{bmatrix} = \frac{333}{EI} ; \begin{bmatrix} a_{00} \end{bmatrix}^{2} = \frac{EI}{333}$$

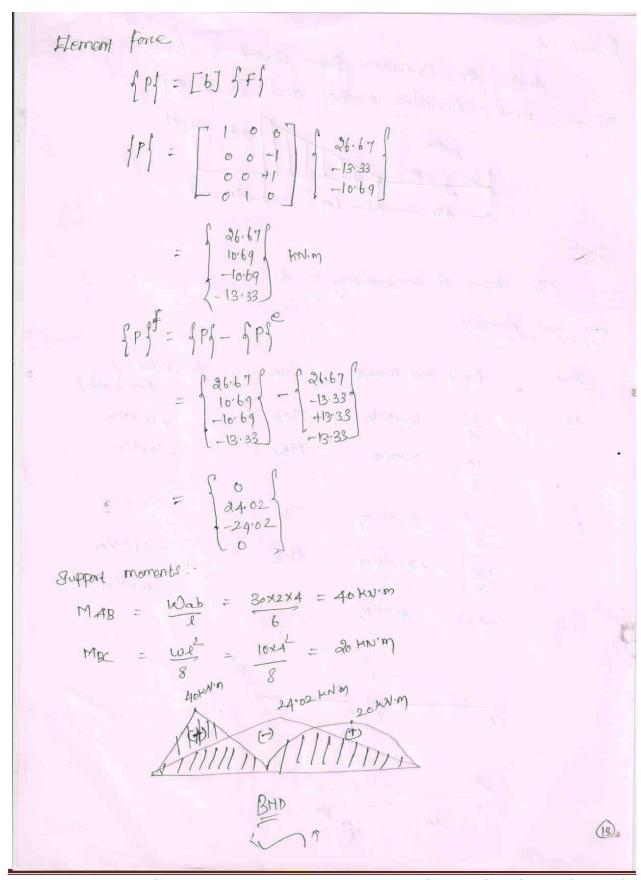
$$\begin{bmatrix} a_{00}^{*} \end{bmatrix} = \begin{bmatrix} b^{*} \end{bmatrix}^{2} \begin{bmatrix} 2 & -1 & 0 & 0 \\ -1 & 2 & 0 & 0 \\ 0 & 0 & 1/33 & -0/47 \\ 0 & 0 & -0/47 & 1/33 \end{bmatrix}$$

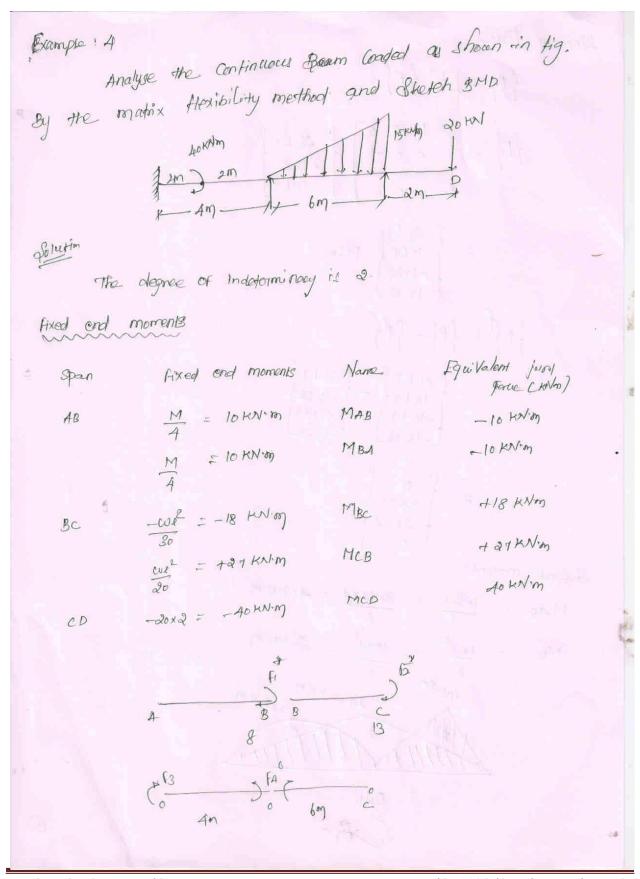
$$= \frac{1}{EI} \begin{bmatrix} 1 - 2 & 1/33 & -0/67 \end{bmatrix} \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 0 \end{bmatrix}$$

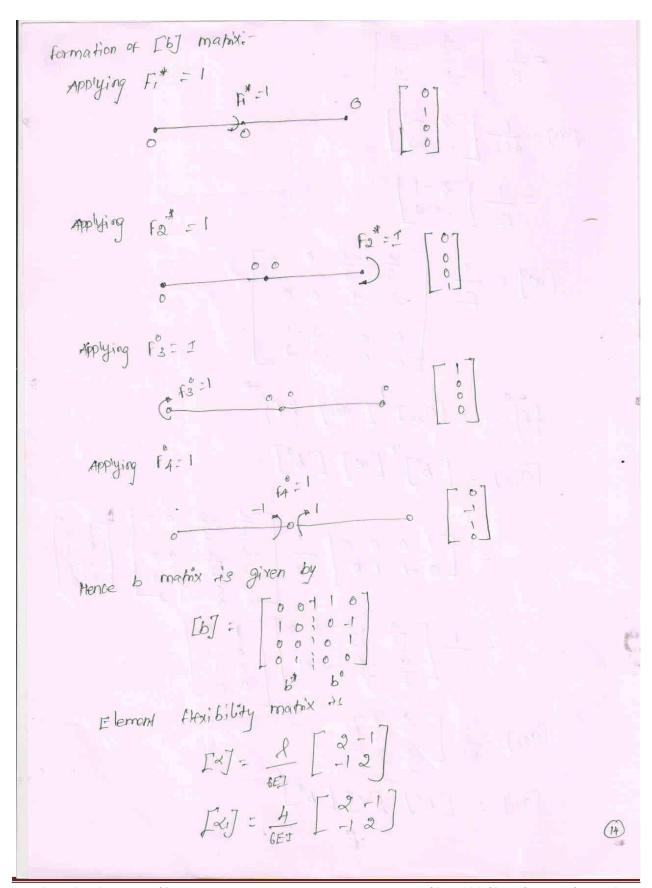
$$= \frac{1}{EI} \begin{bmatrix} 1 - 0/67 \end{bmatrix}$$

$$\therefore \begin{bmatrix} 1 \\ 0 \end{bmatrix} = -\begin{bmatrix} EI \\ 3.33 \end{bmatrix} \times \frac{1}{E2} \begin{bmatrix} 1 - 0/67 \end{bmatrix} \begin{bmatrix} 3/617 \\ -1933 \end{bmatrix}$$

$$\begin{cases} ff_{10} = -10.49 \text{ kN-m} \end{cases}$$







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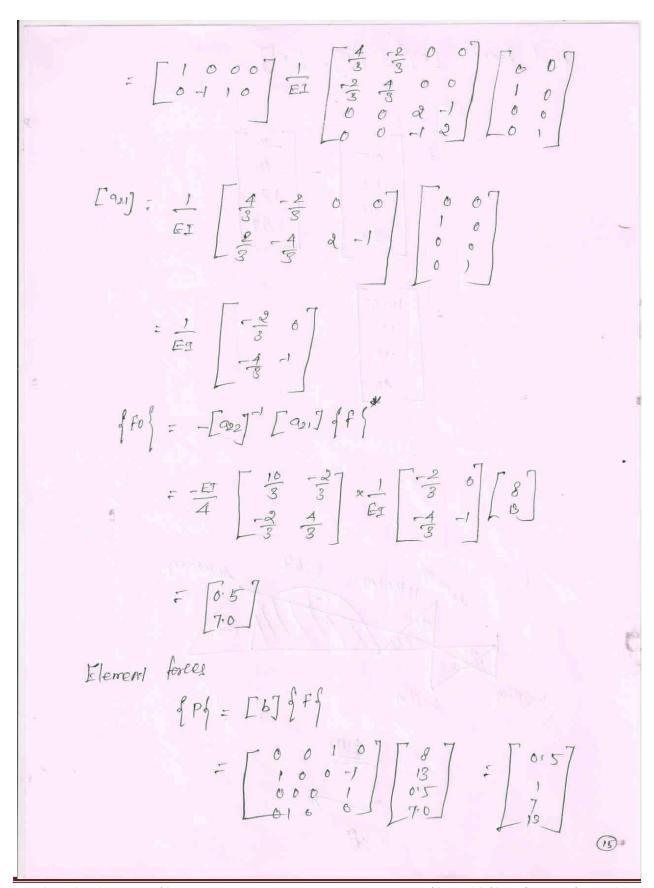
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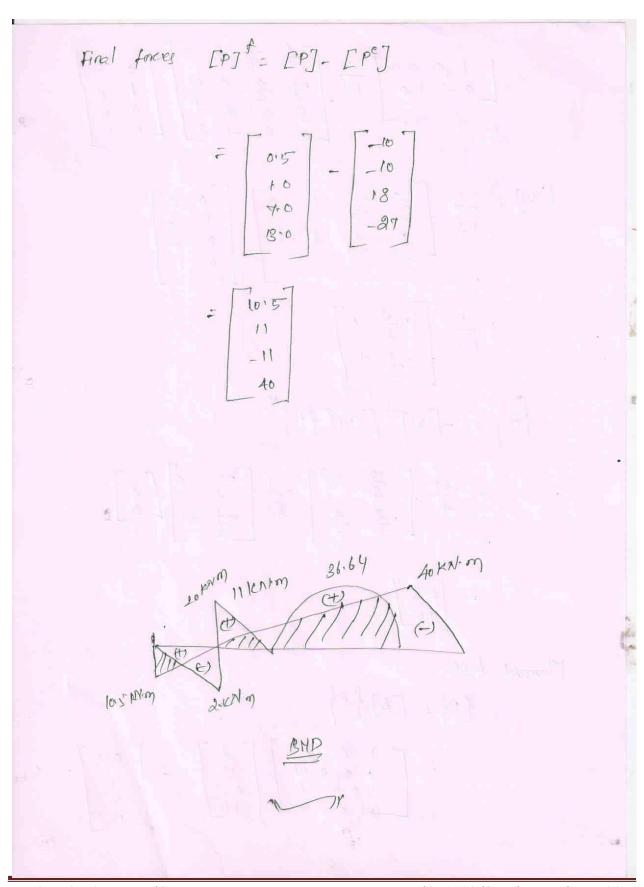
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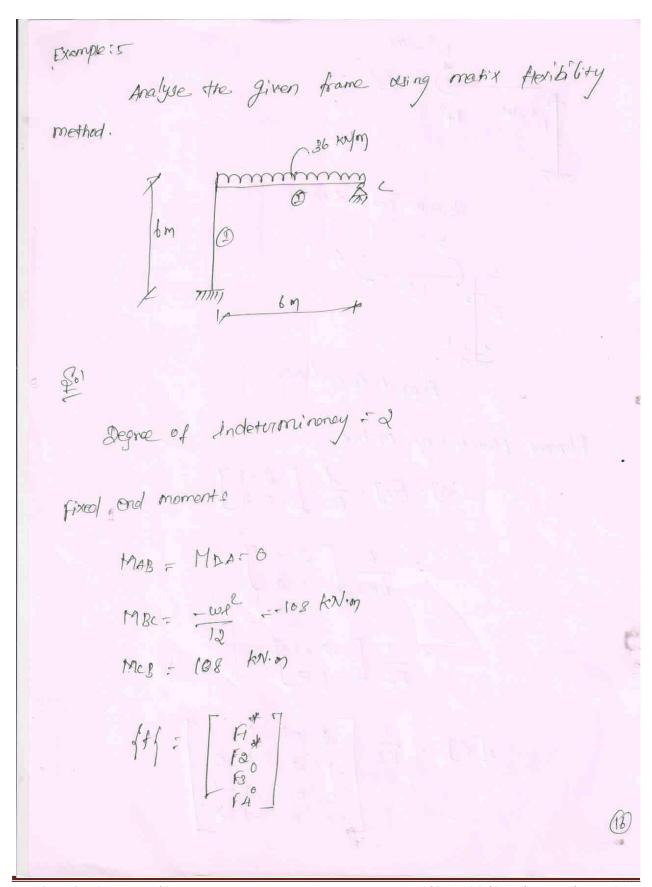
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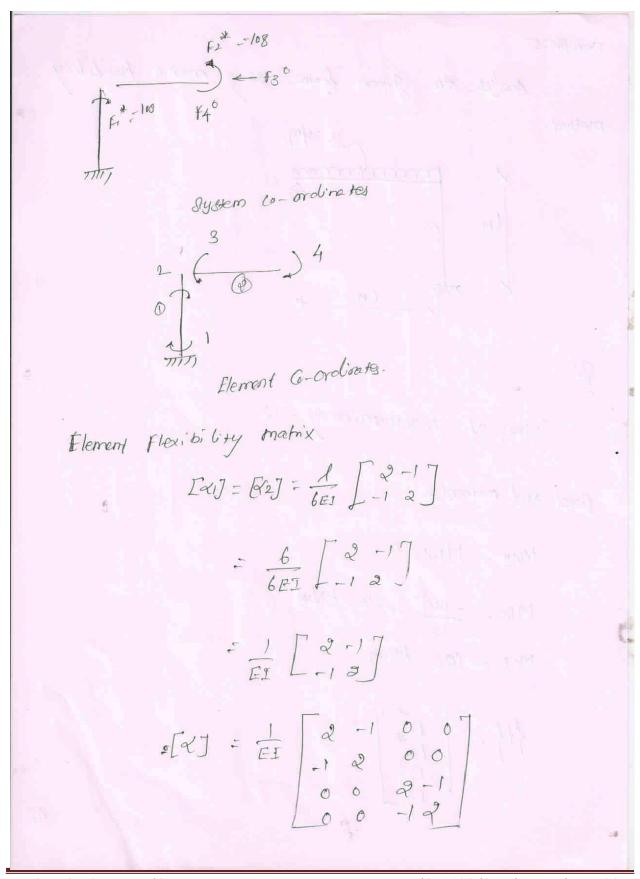
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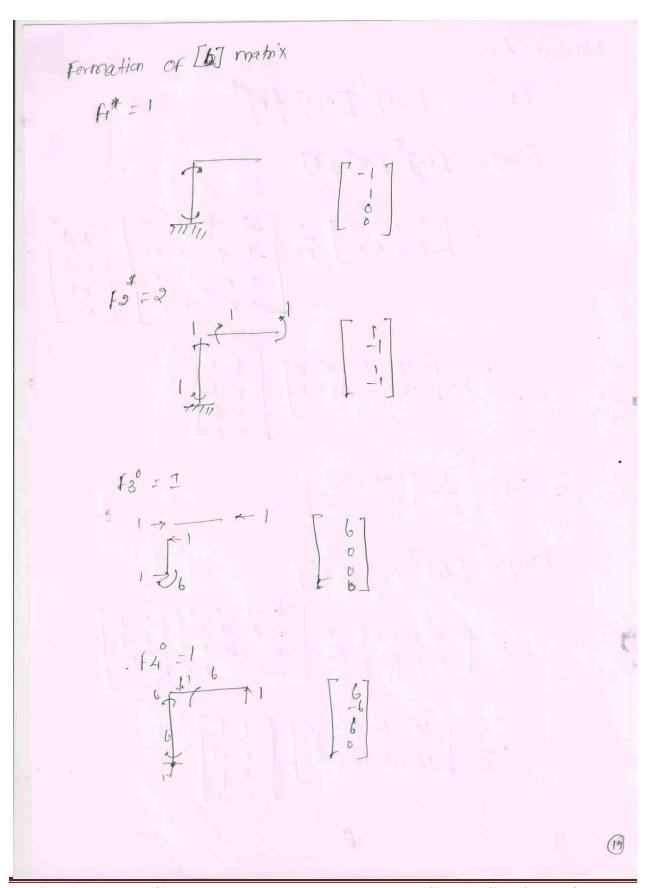
$$\begin{bmatrix}
A_{2} & -A_{3} \\
-A_{3} & -A_{3}$$











Reductant form

$$\begin{cases}
f_{1}^{2} = - [922] [92] f_{1}^{2}
\end{cases}$$

$$[922] = [60] [2] [6]$$

$$= [6000] [1] [200]$$

$$[923] = [1206]$$

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$$[92] = \frac{1}{E1} \begin{bmatrix} -18 & 18 \\ -36 & 54 \end{bmatrix}$$

$$[108]$$

$$= -E1 \begin{bmatrix} 72 & 108 \\ 108 & 288 \end{bmatrix} \begin{bmatrix} 1 \\ E_1 \end{bmatrix} \begin{bmatrix} -18 & 18 \\ -36 & 54 \end{bmatrix} \begin{bmatrix} 108 \\ 108 \end{bmatrix}$$

$$= \frac{1}{9042} \begin{bmatrix} 288 & -108 \\ -108 & 72 \end{bmatrix} \begin{bmatrix} 1944 \end{bmatrix}$$

$$[166] = \begin{bmatrix} 33.14 \\ -15.43 \end{bmatrix}$$
Element force:
$$[P4] = \begin{bmatrix} 15 \end{bmatrix} [166]$$

$$= \begin{bmatrix} 1 & 1 & 6 \\ 1 & -1 & 0 & -6 \\ 0 & 1 & 0 & 6 \end{bmatrix} \begin{bmatrix} 108 \\ 108 \\ 23.14 \\ -15.43 \end{bmatrix}$$

$$= \begin{bmatrix} 46.26 \\ 92.58 \\ 15.24 \\ -108 \end{bmatrix}$$

